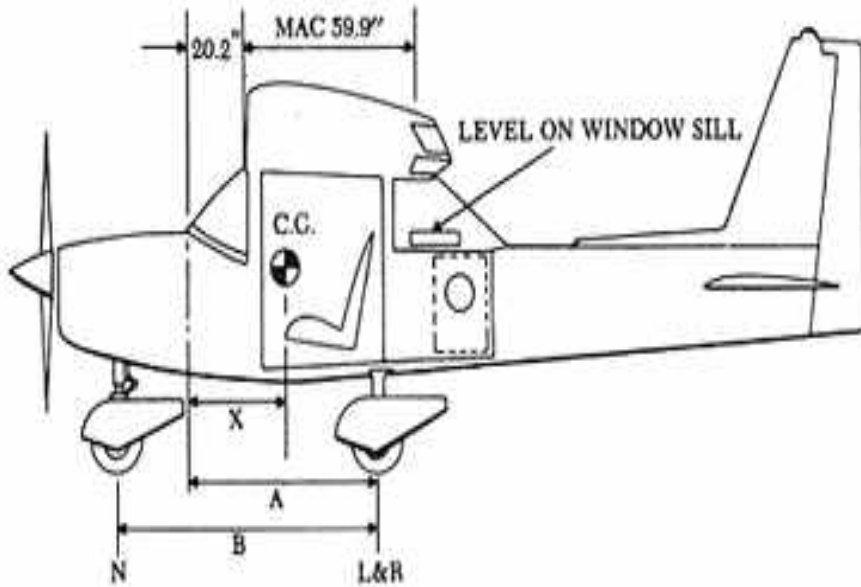


WEIGHT AND BALANCE DATA

DATUM (FIREWALL, FRONT FACE)
STA 0.0



SCALE POSITION	SCALE READING	TARE	SYMBOL	NET WEIGHT
LEFT WHEEL			L	
RIGHT WHEEL			R	
NOSE WHEEL			N	
AIRCRAFT EMPTY WEIGHT (AS WEIGHED)			W	

$$X = \text{ARM (C.G.)} = (A) - \frac{(N) \times (B)}{W}$$

$$X = () - \frac{() \times ()}{()} = () \text{ IN.}$$

FIGURE 20. Typical weight and balance data.